

# The 1983 Chieftain







APPROXIMATE CONTROL SPEED  
MAX. ST. CL. 1400 FT. 100 KIAS  
MINIMUM MANOEUVREING SPEED 100 KIAS  
SEE PILOT'S OPERATING HANDBOOK FOR ADDITIONAL SPEEDS

THIS WARNING MUST BE OBSERVED AS A NORMAL TAKEOFF  
APPROXIMATE CONTROL SPEED  
MAX. ST. CL. 1400 FT. 100 KIAS  
MINIMUM MANOEUVREING SPEED 100 KIAS  
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EGT  
FUEL LOW  
RPM  
MAGNET PRE-ADVICE

110.15 2 1  
286.0 0.0  
12 0.00  
N81350

134.60  
124.90  
108.00  
126.10  
121.30  
117.30  
GROUND CLEARANCE

200 AIRSPEED  
ALT  
CLIMB  
PWR. AIR  
2 MIN. TURN

PHONE  
MAD  
BLAY  
FLST  
OVRD  
CLASH

UP  
LANDING GEAR  
DOWN  
DOWN LOCKED

MAX  
WPAO  
FLAP  
OFF  
ON

FLAP  
SVB TEST

CABIN  
PULL-OUT

WIRE  
PHO

# Piper Aircraft Corporation

1983

Specifications/Performance  
Standard Equipment

## Chieftain

PA 31-350



## Specifications

### ENGINES

Manufacturer	Lycoming
Model	TIO-540-J2BD & LTIO-540-J2BD
Rating (hp @ rpm) Maximum Normal	
Operating Power (MNOP)	315 BHP @ 2400 RPM
Rating (HP @ RPM) Maximum Continuous	
Power (MCP)	350 BHP @ 2575 RPM
Recommended TBO (hrs.)	1600

### PROPELLERS

Manufacturer	Hartzell
Number of Blades	3
Type	Constant Speed/Full Feathering
Diameter (in./cm)	80/203

### WEIGHTS

Maximum Ramp Weight (lbs./kg)	7045/3195
Maximum Take-off Weight (lbs./kg)	7000/3175
Maximum Landing Weight (lbs./kg)	7000/3175
Standard Empty Weight (lbs./kg)	4221/1915
Standard Useful Load (lbs./kg)	2824/1280

### WING AREA AND LOADINGS

Wing Area (ft. <sup>2</sup> /m <sup>2</sup> )	229/21.3
Wing Loading (lbs./ft. <sup>2</sup> )/(kg/m <sup>2</sup> )	30.6/138.6
Power Loading (lbs./hp)/(kg/hp)	10.0/4.5

## Performance

### MAXIMUM SPEED (MNOP) (kts.)/(km/h)

(TAS at Average Cruise Weight)	231/427
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### CRUISING SPEEDS

(TAS at Average Cruise Weight)

Power %	Cruise Altitude (ft./m)	Cruise Speed (kts.)/(km/h)
75	20,000/6096	221/410
75	12,000/3658	205/380
65	20,000/6096	210/389
65	12,000/3658	191/354
55	15,000/4572	177/328
55	12,000/3658	173/321

### CRUISE RANGE

Includes Allowance for Fuel Used During Start, Taxi, Take-off, Climb and a 45 Minute Reserve at Long Range Cruise Power.

Power %	Cruise Altitude (ft./m)	Standard Fuel	Optional Fuel
75	20,000/6096	885/1640	1210/2241
75	12,000/3658	855/1584	1100/2148
65	20,000/6096	925/1714	1260/2333
65	12,000/3658	900/1668	1225/2267
55	15,000/4572	950/1760	1290/2389
55	12,000/3658	950/1760	1290/2389

### DIMENSIONS

Wing Span (ft./m)	40.67/12.4
Length (ft./m)	34.63/10.6
Height (ft./m)	13.0/4.0
Cabin Length (in./cm)	151/384
Cabin Width (in./cm)	50/127
Cabin Height (in./cm)	51.5/131
Passenger door size (in./cm)	45 x 27.5/114 x 70
With adjacent cargo door (in./cm)	45 x 44.5/114 x 113
Forward luggage door size (in./cm)	25 x 28/64 x 71
Nacelle locker door size (in./cm)	20 x 40/51 x 102

### USABLE FUEL

Standard fuel (gal./L)	182/689
Optional fuel (gal./L)	236/893

### OIL CAPACITY (gal./eng.)/(L/eng.)

3/11.4

### BAGGAGE

Luggage capacity (lbs./kg)	
Nose	200/91
Aft	200/91
Nacelle (two)	300/136
Nacelle (two) w/opt. fuel	100/45
Luggage space (ft. <sup>3</sup> /m <sup>3</sup> )	
Nose	14/.40
Aft	22/.62
Nacelle (two)	26.5/.75
Nacelle (two) w/opt. fuel	12.0/.34
Cargo space (ft. <sup>3</sup> /m <sup>3</sup> )	
Total	217/6.15
Total w/opt. fuel	202.5/5.73

### RATE OF CLIMB

(At Sea Level and Gross Weight)

Two Engines (MNOP) - (fpm/mpm)	1120/342
Single Engine (MCP) - (fpm/mpm)	230/70

### SERVICE CEILING

Two Engines (MNOP) (ft./m)	*24,000+/7317+
Single Engine (MCP) (ft./m)	13,700/4176

### STALL SPEEDS

Power off, Flaps Down (kts./kmh) IAS	74/137
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### TAKE-OFF DISTANCE

(Sea Level, Zero Wind, Standard Temperature)

Ground Run (ft./m)	1350/411
Total distance over 50 ft. obstacle (ft./m)	2510/765

### LANDING DISTANCE

(Sea Level, Zero Wind, Standard Temperature)

Ground Roll (ft./m)	1045/319
Total distance over 50 ft. obstacle (ft./m)	1880/573

\* 24,000 feet is maximum approved altitude for the Chieftain

# Standard Equipment

## POWER PLANTS AND PROPELLERS

Engines — 2 counter rotating turbocharged  
Lycoming, with Bendix fuel injection; 350 hp at  
2575 rpm; direct drive, 6 cylinder, dual ignition with  
shielded ignition harness  
Dynafoal engine mounts  
Propellers — 2 Hartzell, 3 blade extended hub,  
constant speed, full feathering, with propeller  
spinners  
Propeller governors, two  
Geared starters, 24 volt, two  
Air filters, two  
Oil coolers with thermostatic control, two  
Oil filters, two  
Alternate air sources — automatic, with manual  
controls, two  
Electric cowl flaps, two  
Dry pneumatic pumps, two  
Engine driven hydraulic pumps, two

## FLIGHT INSTRUMENTS AND INDICATORS

Piper Truspeed Indicator  
Magnetic compass (illuminated)  
Sensitive altimeter (In. and Mb.)  
Piper pictorial turn rate indicator  
Rate of climb indicator  
8-day clock  
3" pictorial gyro horizon (Air Driven) TSO'd  
3" directional gyro (Air Driven) TSO'd  
Gyro air filter  
Outside air temperature gauge  
Ammeter  
Annunciator panel:  
Alternator inoperative L/R  
Pneumatic source inoperative L/R  
Low fuel flow L/R  
Fuel boost pump inoperative L/R  
Heater over temperature  
Flap  
Cabin door ajar  
Nose baggage door ajar  
Dual manifold pressure gauge  
Pneumatic gauge  
Dual tachometer  
Flight hour recorder  
Fuel quantity gauges, two  
Dual fuel flow gauge  
Dual fuel pressure gauge  
Combination oil pressure, oil temperature and  
cylinder head temperature gauges, two  
Dual exhaust gas temperature gauge  
Aileron trim position indicator (illuminated)  
Elevator trim position indicator (illuminated)  
Rudder trim position indicator (illuminated)  
Wing flap position indicator (illuminated)  
Cowl flap position indicator (illuminated)

## COCKPIT, FLIGHT AND GROUND CONTROLS

Flight primary — dual with ram's horn type wheels.  
Provision for elevator trim switch, mike button, pitch  
sync. and autopilot disconnect on pilot and copilot  
control wheels.  
Flight trim — pedestal (illuminated)  
Aileron  
Rudder  
Elevator

## Engine controls:

Throttle, two  
Propeller, two  
Mixture, two  
Cowl flap (electric), two  
Alternate air, two  
Engine controls' friction locks  
Dual flight controls  
Stall warning horn  
Cockpit and cabin heater/ventilation master controls  
Steerable nose wheel  
Brakes  
Pilot's toe brakes  
Provisions for copilot's toe brakes  
Parking brake  
Landing gear, retractable hydraulic  
Landing gear actuator control  
Landing gear warning horn  
Landing gear emergency extension — manual  
hydraulic  
Wing flaps, 0° to 40° electrically operated — includes  
proportional preselect feature  
Wing flap position indicator  
Fuel control pedestal  
Fuel tank selectors, two  
Crossfeed selector  
Shut-off valves/with warning lights  
Cabin exhaust vent  
Alternate instrument static source and control

## ELECTRICAL PROVISIONS

Dual 28 volt, 70 amp alternators  
24 volt, 17 amp hour battery  
Dual paralleling voltage regulators with overvoltage  
relays  
Resettable type circuit breakers  
Ammeter  
External power supply receptacle  
Circuit breaker panel, pilot — essential buss  
Circuit breaker panel, copilot — avionics

## AVIONICS PROVISIONS

Cabin speaker  
Cockpit speakers, two  
Headphone and microphone jacks — dual  
External avionics racks and cabling provisions  
Provisions for automatic locator beacon  
Wide choice of optional avionics available  
Circuit breaker panel, copilot — avionics

## FUEL SYSTEM

Four bladder cell type fuel tanks with 192 gallon total  
capacity, 182 usable, equipped with NACA type  
anti-icing non-siphoning vents w/main tank baffles  
Engine driven fuel pumps, two  
Electric auxiliary fuel pumps, two  
In-line low pressure fuel pumps, two  
Fuel filters with quick drains, two  
Fuel tank sump quick drains, four  
Crossfeed drain  
Fuel shut-off valves on engine firewall, two

## ICE PROTECTION PROVISIONS

Heated pitot head — pilot's  
Elevator horn anti icing boots  
Deicing group available for flight in icing conditions

## LIGHTING PROVISIONS

External lights  
Anti-collision strobe lights, three  
Navigation lights, three  
Landing/taxi lights, two  
Courtesy lights  
Nose luggage compartment  
Stair door and rear luggage compartment  
Crew area  
Cockpit lights  
Landing gear position, four:  
down/locked, three; intransit/not locked, one  
Instrument panel, switch and circuit breaker panel  
lighting, rheostat controlled  
Instrument panel back-up lights, two  
Overhead engine switch panel lighting, rheostat  
controlled  
Overhead map lights, two (white)  
Fuel control pedestal light (white)  
All lighting rheostats — centrally located in  
overhead panel  
Cabin lights  
Passenger reading lights, individual, five  
Rear dome light

## CABIN COMFORT SYSTEM

Janitrol 50,000 BTU combustion heater,  
thermostatically controlled with baseboard cabin  
outlets  
Windshield defrosters  
Sidewall cabin fresh air vents, individually controlled,  
eight  
Recirculation/fresh air blowers with HI/LO speeds  
Provisions for air conditioning  
Cabin exhaust vent

## EXTERNAL FEATURES

Three tone exterior paint design, in a wide choice of  
color combinations  
Polyurethane paint — exterior finish  
Corrosion proofing — internal and external  
Main wheels — 6:50 x 10 with disc brakes; tires with  
tubes — 6:50 x 10, 8 ply rating  
Nose wheel — 6:00 x 6, tire with tube — 6:00 x 6, 6 ply  
rating  
Aircraft brakes  
Stowable towbar  
Tie down rings, three  
Jack pads  
Bonding straps across all control surfaces and  
fiberglass parts for lightning strike protection  
Nose gear safety mirror  
Cabin entrance door with built-in steps and  
pneumatic extender  
Meets FAR Part 36 noise requirements  
Cabin, nose and nacelle luggage door locks with keys  
and carpeting

## COCKPIT AND CABIN APPOINTMENTS AND PROVISIONS

Choice of eleven interior color themes, which  
includes:  
fabric and vinyl seats, fabric side panels, wall to  
wall carpeting, vinyl headliner and color keyed  
curtains

*Continued on back page*

# Standard Equipment Continued

Pilot/copilot seats — fabric and vinyl with headrests, folding armrests and oxygen mask storage underneath each seat. Seats adjust fore and aft, vertically and tilting with shoulder and safety belts and inertia reels.

Flashlight that stows under pilot's seat

Ash trays, two in crew area

Storm windows — pilot and copilot

No smoking/seat belt lighted signs with cockpit control switches

Forward cabin divider curtain

Cigarette lighter, cockpit

Scuff plates, pilot and copilot

Shock mounted instrument panels

Removable instrument panels, three

Two-piece windshield

Sun visors, two

Four reclining and adjustable passenger seats — fabric and vinyl in Club arrangement with headrests, folding armrests, seat belts, oxygen mask storage underneath each seat, and magazine storage pockets on the back of each seat

Provisions for up to eight adjustable passenger seats

Ash trays, four located in each seat outboard armrest

Emergency exit window

Double glazed windows

Quietized soundproofing

Window curtains

Coat hanger support bar

Coat hangers, six

Luggage compartments with security straps:

Cabin walk-in, 22 cu. ft. — 200 lbs.

Fuselage nose, 14 cu. ft. — 200 lbs.

Nacelle, 13.25 cu. ft. — 150 lbs. each side

Removable floorboards

Provision for oxygen installation

Compass card

Compass card holder

Weight and balance plotter

Pilot's Operating Handbook — Jepp size

Passenger briefing cards

Aircraft logbook

Engine logbooks

Certificate of Airworthiness

## PRODUCT SUPPORT

Piper Warranty Form

Piper Service Center Directory

Inspection Forms

The performance information is based on an airplane flown at gross weight under standard sea level atmospheric conditions except as noted and based on the latest data available at the time of publication approval. Take-off and landing performance is optimum. Actual performance depends on pilot techniques, operating surfaces and other factors. It is the responsibility of the pilot to determine that all operations are conducted within approved limits of design gross weight, center of gravity, and in accordance with the FAA-approved Airplane Flight Manual which is the only official source of operating parameters and performance information.

In accordance with GAMA format, range provides for taxi, take-off, climb at MCP, cruise at stated mixture and descent with 45-minute reserve at maximum range power. Empty weight includes unusable fuel, full operating fluids and full oil.

Piper Aircraft Corporation reserves the right to make changes in specifications, materials, equipment or prices at any time without prior notice or to discontinue models as required.

Your Piper Dealer has listings of a wide variety of optional equipment and avionics. Items most frequently chosen by owners are packaged for factory installation at substantial price savings.



**PIPER AIRCRAFT CORPORATION**  
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